

**Serial No. Not Yet Assigned**  
**Atty. Doc. No. 2002P84041WOUS**

Amendments To The Abstract:

In the English translation document, please add the section heading at page 8 line 1, as follows:

--ABSTRACT--

Please add the paragraph in the English translation document at page 8 line 1, after the newly added ABSTRACT, as follows:

A turbine blade or vane comprises an impingement tube located generally in a radial direction within the blade or vane aerofoil, the impingement tube comprising two parts, extending into the blade or vane from opposite radial ends thereof and locating against a specially formed rib which extends generally chordwise around the leading edge of the aerofoil.